Getting started with TetrUSS on Mac OS X

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This document is intended to provide a brief introduction to using TetrUSS CFD software on Mac OS X. It is essentially a self guided "tour", designed to familiarize users with TetrUSS. This document is not intended to serve as training or documentation. For detailed instruction in TetrUSS, users are encouraged to attend the TetrUSS training course offered monthly at NASA Langley Research Center. See the document "Training and Support" for more details.

Welcome to TetrUSS on Mac OS X! TetrUSS is one of the most advanced CFD tools available, capable of numerous applications in aerodynamics and fluid dynamics. Before getting started with our tour, let's discuss some of the software that will be used here:



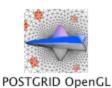
GridToolCocoa

GridToolCocoa is used to prepare geometries for grid generation. Typically, a user will import a geometry (IGES, Plot3D, or curves) into GridToolCocoa, and then lay the foundation for grid generation using VGRID OpenGL.



VGRID OpenGL

VGRID OpenGL generates unstructured triangular surface meshes and unstructured tetrahedral volume grids using the advancing layer and advancing front methods.



POSTGRID OpenGL completes and repairs grids created by VGRID OpenGL. In addition, POSTGRID OpenGL can be used for grid movement and adaptation.



Preface

Preface is a utility preprocessor that prepares grids for use in the USM3D flow solver. Among other things, Preface computes the minimum distance between cells and viscous surfaces, for use in turbulence modeling.



3D Euler and Navier-Stokes flow solver. USM3D is geared towards aerodynamic simulations, but also has a wide range of applications in fluid dynamics.



histPLOT

Convergence tracking and plotting utility. histPLOT makes it easy to follow a solution and see how it is progressing, by quickly plotting out residual along with selected forces and moments.



Utility that converts USM3D solution files into ASCII or binary data files in a Tecplot-compatible format.



Tecplot 9.2

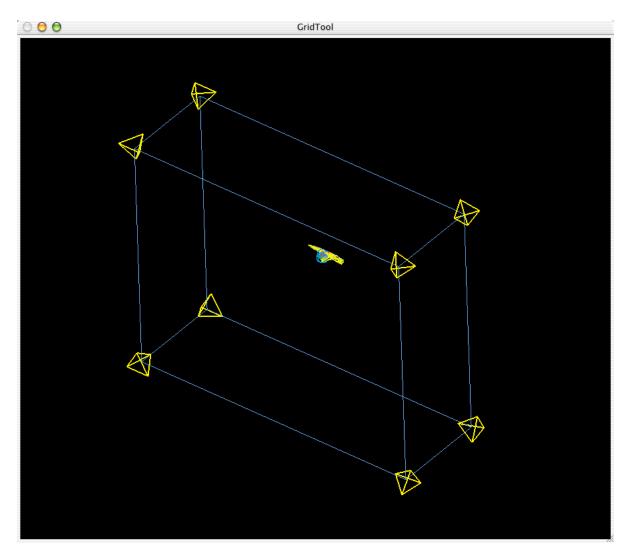
Amtec Tecplot 9.2 (optional commercial software) is a 2D and 3D plotting and data visualization tool that can handle large CFD data sets. It is used to explore, visualize, and present USM3D results.

In this introduction, we will walk through the steps of applying TetrUSS to simulate transonic cruise of a generic "advanced technology fighter" (ATF) aircraft configuration, going from geometry preparation to grid generation to running the flow solver to postprocessing and visualizing results. We are going to start from a pre-defined geometry contained in the GridTool restart file "atf.rst" located in the "Sample Project" folder included with the TetrUSS software. It's a good idea to make an "ATF" folder somewhere on your hard drive, and copy the restart file into that folder (we'll write subsequent files into the same location). While you're at it, copy the "atf.inpt" file into your ATF folder as well. Keep the original files in a safe place, in case you need to recover them.

To follow the grid generation portion of the tour, at least 80MB of free RAM is required. To run the USM3D flow solver, approximately 480MB of free RAM is required.

GridToolCocoa

To start off, launch GridToolCocoa from the finder by double-clicking its icon. Go to the "File" menu, choose "Open", and navigate your way to the "atf.rst" file in your ATF folder. Open the file. GridToolCocoa will read in the geometry data, and you'll see it appear in the main graphics window (which can be resized to your liking). Several panels will pop up as well. Go to the "Viewing Controls" panel and check the box for "Plot3D Mouse". Now, place your mouse pointer in the graphics window. Combined with movement, the mouse buttons do the following: left=rotate(x)/rotate(z), middle=zoom/rotate(y), right=translate. If you do not have a 3-button mouse, control-click emulates the middle button and option-click emulates the right button. You should be able to explore the geometry by mousing around (click "Reset" on the Viewing Controls panel if you get lost). What you'll see is a model of half the ATF configuration, mounted to the X-Z "symmetry" plane inside a box (the box is the flow domain).



The GridToolCocoa graphics window

You may notice several things in the GridTool model; first, a wireframe of the ATF. This was constructed from a CAD definition of the ATF (originally in the form of an IGES file). To see the CAD surfaces, you can bring the "Surface Attributes" panel to the front (also accessible from the "Window" menu). Use the popup menus in this panel to change the surface attributes; change the "Selection" popup menu to "All" and "Display" popup menu to "Hidden". To improve graphics performance while moving the ATF around with surfaces on, hold down the shift key. Note that you can change the "Display" popup menu to "Off" to turn off the surfaces.

Another thing you may notice are yellow (or purple when active) grid "sources". These are user-specified, and intended to govern the grid size and spacing during subsequent grid generation with VGRID. There are small sources near the fuselage chine, at the wing tip, and along the leading and trailing edges. Large sources are placed near the outer corners of the domain. A final thing you may notice are "patches", shown as blue objects on the ATF surface. Active patches have arrows on their boundary, and a normal vector in the center of the patch. Like sources, patches are user-defined. These are setup to strategically divide the CAD geometry into smaller, geometrically simpler surface components.

This particular example has already been through many hours of geometry preparation, where a NASA engineer has taken the raw IGES geometry, patched it up, and specified sources. This is the most labor-intensive part of using TetrUSS, and will not be covered here. This process is part of the TetrUSS training course, however.

OK, now we're ready to move onto grid generation. Go up to the File menu and choose "Save". If needed, navigate into your ATF folder. At the bottom of the save pane, change the file type to "d3m (VGRID)" (it's the last option in the popup menu). Click "Save". GridToolCocoa should have created several new files in your demo folder: atf.d3m, atf.igs, and atf.mapbc.

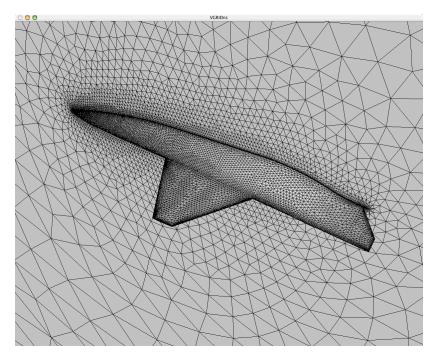
Grid Generation with VGRID OpenGL

Head back to the Finder. Open both the ATF folder (with the new file icons showing) and the TetrUSS folder (i.e., /Applications/TetrUSS). Drag the icon from the "atf.d3m" file onto the "VGRID OpenGL" application icon and drop it. Through the magic of AppleScript, a Terminal window will open up (Terminal.app will launch if needed) and VGRID will start a graphics window. If possible, position your windows so that you can see both the terminal window and the graphics window (or part of each one). Click on the title bar of the graphics window to bring it to the front. When you place your mouse in the window, you should see the cursor turn into a "?". This the GLUT (part of OpenGL) interface telling you it's time to right-click the mouse; do so now. Select "Viscous" from the popup menu.

Next, you should see a wireframe of the ATF geometry and flow domain. Like in GridToolCocoa, the various mouse buttons will allow you to rotate, zoom, and translate. If you get lost, press "r" on the keyboard to reset the view. When you're done looking at the wireframe, press the "esc" key on the keyboard (i.e., the ESCAPE key in the upper left corner). Notice another "?" pops up, so right-click. Tell VGRID to display background grid sources in the popup menu. It will display the sources specified in GridToolCocoa. Hit "esc" again, and then right-click the mouse and tell VGRID to display points along patch boundaries (this shows the approximate surface grid distribution along patch boundaries).

When you're done looking at the points, hit "esc" one more time, right-click, and tell VGRID to perform surface triangulation without display. Action should switch to the terminal, and VGRID will start computing the surface mesh. When complete, the graphics window will come to the front. You can use the mouse to explore the surface mesh (start by zooming in). You'll see the surface mesh of the ATF and the boundaries of the flow domain. For easier control, place the mouse arrow over the ATF fuselage near the symmetry plane, and push the "a" key on the keyboard to set the center of rotation at that point. You can toggle the "t" key to switch between the mesh, the wireframe, and the shaded surface.

When you're done looking at the surface mesh, press "esc", right-click, and select "Exit plotting" from the popup menu. Right-click again and select "Generate volume grid and show ▶ every 1000 iterations". VGRID will begin generating the viscous (boundary layer) grid on the ATF (you'll see it growing in layers). This will happen quickly on a fast Mac. On a slow Mac, you can press "s" in the graphics window to toggle graphics off, which will make the grid generation go faster (hit "s" again to toggle graphics back on).

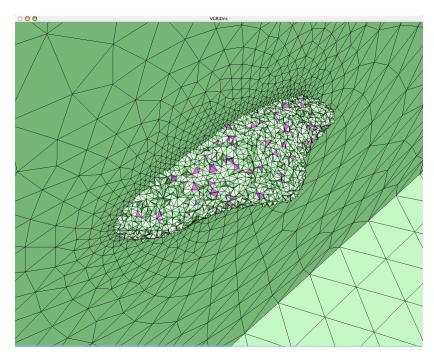


ATF surface mesh in VGRID

When the viscous grid generation is complete, the terminal will display "No negative cells found". Go to the VGRID graphics window, hit "esc", right-click, and tell VGRID to save the viscous grid and stop. VGRID will quit (you can safely ignore any warning messages VGRID may display in the terminal after it quits).

Now, we need to run VGRID again to generate the volume grid. You could do the drag and drop trick again, but at this point it's easier to go back to the terminal window (the one that opened up before) and type "vgrid -i atf" (do not use quotes), followed by a return. VGRID will launch again. Same routine as before: right-click in the graphics window, and select "viscous". You can zoom into the wireframe and position the ATF in your view. Now hit "esc" and you'll see the ATF and all the viscous layers. You can zoom in and explore as desired.

When you're done looking around, press "esc", right-click, and select "Exit Plotting". Right-click again, and choose "Generate volume grid and show • every 100 iterations". VGRID will begin growing the volume grid, starting with the smallest cells (near the wing tip – zoom in for a look). The volume grid generation process will take several minutes, even on a fast Mac. As before, you can use "s" to toggle graphics off in order to improve performance (though it's fun to leave graphics on and watch VGRID grow the grid). VGRID will generate grid until the whole domain is filled, and then automatically quit.



Volume grid generation in VGRID

POSTGRID OpenGL

At this point, we need to repair any problems with the grid. For this, POSTGRID OpenGL is used. Back in the terminal, type "postgrid -i atf" and hit return. Bring the graphics window to the front, right-click, and select "viscous". Right-click again and select "No grid movement", and right-click once more and select "Unstructured grid (.cogsg)". When the wireframe appears, re-orient your view to zoom in on the ATF. Press "esc", and right-click. Choose "Remove 1 layer of cells (viscous and inviscid portions)" from the popup menu. Right click again and select "Display grid pockets". POSTGRID will show pockets in the grid which must be repaired. When you're done looking at the pockets, press "esc", right-click, and select "Yes, generate volume grid on the fly \(\right\) every 100 iterations".

POSTGRID will attempt to close the pockets, and this generally takes several iterations. In this case, one more iteration is required. So, right-click, select "Remove 1 layer of cells (inviscid portion only)". Right-click again, and select "Display grid pockets". Press "esc", right-click, and select "Yes, generate volume grid on the fly ▶ every 100 iterations".

After this last iteration, the message "GRID IS COMPLETE!" should appear in the terminal. Now, you can right-click in the graphics window and select "Write new data > Unformatted (.cogsg) file". When file I/O is complete, hit "esc", right-click, and select "Quit". You will need to right-click once more and select "Yes, quit".

Preparing and Running USM3D

Now, the grid generation is complete, and we can prepare to run the USM3D flow solver. Step one is to run "Preface", which is a preprocessor. To do this from the command line, just type "preface atf" and hit return (you could also drag and drop any one of the "atf.*" files onto the "Preface" icon in the finder). Wait about a minute while preface runs in the background.

When preface has finished, you can run the USM3D flow solver (to actually simulate flow over the ATF configuration). To start off, type "usm3d.r4 -c atf" in the terminal and hit return. This should indicate that all required files exist. Notice that the grid contains about 662,000 cells according to USM3D. Now, type "usm3d.r4 -s atf" in the terminal and hit return. USM3D will tell you how much memory is required to run the CFD simulation; this should be a bit less than 480MB (for single precision, USM3D requires about 720 bytes per cell, and 662,000 x 720 works out to a bit less than 480 MB).

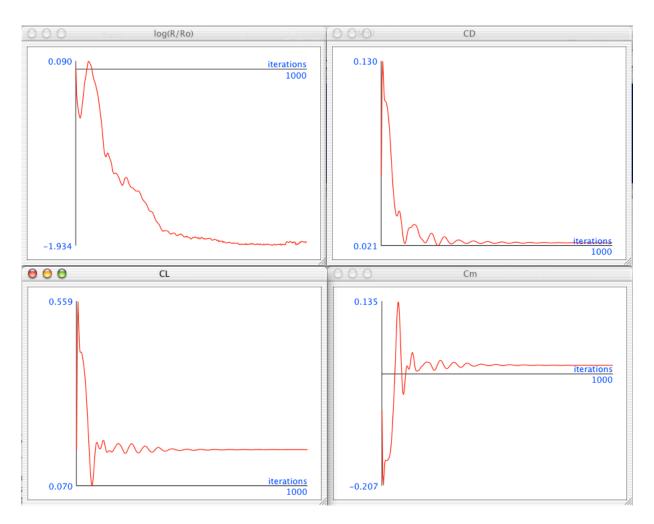
Important: Don't go any further unless you have 480MB of free RAM!

Once you're ready to go, type "usm3d.r4 atf" in the terminal and hit return. USM3D will begin to run. The "atf.inpt" file is set up to run 5 iterations of the flow solver. This will take several minutes, even on a fast Mac. If you want to obtain a converged solution for this case, USM3D must be run for 1000 iterations (this takes about 16 hours using one processor of a dual-1.25GHz PowerMac G4). To do this, edit the "atf.inpt" file and change "ncyc" (in the lower left corner) to 1000. Make sure the file is saved with UNIX carriage returns (using standard UNIX text editors is fine, or something like BBEdit Lite with the UNIX CR option). Type "usm3d.r4 atf" in the terminal and hit return to start the run.

```
ATF CONFIGURATION - Transonic Cruise - USM3D Version 5.2 Control File
Mach
        alpha
                  beta
                           ReUe,mil
                                        Tinf, dR
                                                    itwall
                                                               Tw/Tinf
                                                                          ipwall
0.75
        3.00
                  0.00
                                                    0
                                                                           0
                           0.0058396
                                        460.0
                                                               -1.0
sref
        cref
                  bref
                            xmc
                                       ymc
                                                  zmc
30600
        182.86
                  198.11
                            364.6
                                       0.
                                                  93.1
                            cfl2
                                     cflmin
                                                GS tol
impl
       dt/cfl1
                  iramp
       -5.
                  200
                            100.
                                     5.0
                                                 0.005
irest
        mstage
                  iresmth
                               dqmax
                                         p_break
                                                    p min
                                                              limiter
                               0.25
                                         0.01
                                                    0.001
0
         3
                  1
                                                              1.
nupdate
                       ipltqn
            nwrest
                                   idiagnos
                                               nodeypl
            50
                        2
10
                                    0
                                                0
                     high-bc
                                   ifds
                                             ivisc
                                                       EV lim
iorder
         lapl-avg
         2
                      1
                                    1
                                               3
                                                          0.1
ncyc
         nengines
                       compF&M
                                    cldes
1000
                                     0.
```

If you decide to bail out of USM3D while it's running, just hit control-c in the terminal, and it should interrupt the solution procedure. A more graceful way to stop the run (which will also preserve the solution) is to create a text file called "usm3d.stop" in the ATF directory, and place a "1" (no quotes) on the first line. The text file needs to have UNIX carriage returns. To restart or continue a solution in progress, modifications will need to be made to the atf.inpt file, and these are beyond the scope of this introduction.

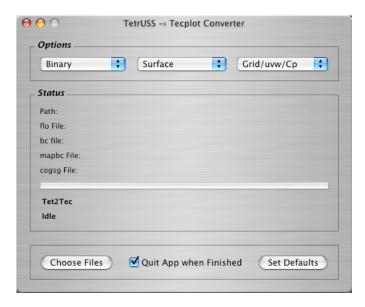
While the solution is running, you can monitor convergence with the "histPLOT" utility. There are two ways to do this: 1) from the Finder, you can go into your ATF folder, grab the icon of the "hist.plt" file, drag it and drop it on the "histPLOT" icon in /Applications/TetrUSS, or 2) open up a terminal window (the original window we were using is probably tied up while USM3D runs), change into the "ATF" directory, and type "histPLOT" at the command line followed by a carriage return. Either way, the AquaTerm terminal plotting utility will launch, and histPLOT will construct several plots tracking convergence. You'll see solution residual (logR/Ro), lift coefficient (CL), drag coefficient (CD), viscous drag coefficient (CDV), pitching moment coefficient (Cm), turbulence residual (logT/To), and CFL number (CFL). Essentially, USM3D needs to run until solution residual has dropped a couple orders of magnitude (each –1 increment is an order of magnitude in the logR/Ro plot) and the aerodynamic coefficients (CL, CD, Cm, CDV) level off. When everything settles out, the solution is said to be "converged". Again, this takes about 1000 iterations for the sample ATF case.



Typical convergence history (residual, CL, CD, Cm) for ATF Case over 1000 iterations

Postprocessing and Visualization

When the solution has finished, integrated forces and moments for the ATF configuration are given near the end of the "tet.out" text file (in the form of aerodynamic coefficients in various axis systems). You can also take a physical look at the solution by visualizing and plotting it in Amtec Engineering's Tecplot 9.2 software for Mac OS X (see www.amtec.com, where a 14-day trial is available). To do this, the solution must first be postprocessed with the "Tet2Tec" utility. Easiest way to run this is from the Finder. Go into /Applications/TetrUSS, and double click the "Tet2Tec" icon. After the utility launches, set the three popup menus (at the top of the Tet2Tec window) as follows: Binary, Surface, Grid/uvw/Cp. This tells Tet2Tec that we want a binary output file of the surface mesh, and we want the data to contain the grid (x,y,z points), velocities (u,v,w in 3D), and pressure coefficient (Cp). There's actually a ton of data in the CFD solution, and we're just extracting a small subset of it to make things go faster. You could extract different data variables, and even look at data in the flowfield around the ATF.



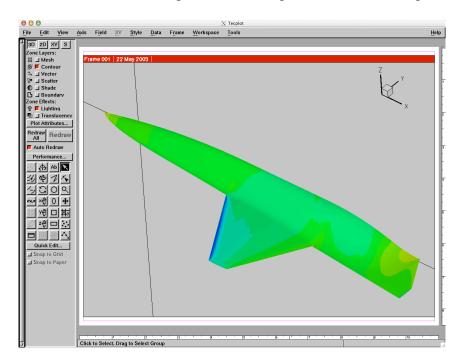
Tet2Tec Utility

Now, click the "Choose Files" button in the Tet2Tec window, and navigate your way to the ATF folder. Once you're there, hold down the command key and click on the following four files to select them all: atf.bc, atf.cogsg, atf.flo, and atf.mapbc (note that Tet2Tec will ignore any other files in the ATF folder). When all four files are highlighted, click the "Choose" button. Tet2Tec will process the files and produce an "atf.plt" file suitable for Tecplot. Tet2Tec will quit automatically if the "Quit app when finished" checkbox is selected.

Note: for Tet2Tec to run properly, Tecplot 9.2 must be installed. Tet2Tec uses the "preplot" utility installed by Tecplot to create .plt binary files. If Tecplot is not installed, you can only use Tet2Tec to write ASCII data files (an option in the left popup menu).

We won't go into specifics about Tecplot here (see Amtec's documentation for more details), other than to discuss the procedure for reading the ATF solution into Tecplot and seeing the results. From within Tecplot, go to the "File" menu, choose "Load data from file(s)", and navigate to the "atf.plt" file you just created. Highlight the file, and click "OK". Once the data has loaded, change the plot type to "3D". You can rotate/translate the data set to get the best view. You should see the unstructured mesh for all surfaces in the domain; the surface of the ATF, the symmetry plane, and the outer "box".

If desired, go to the "Field" menu, choose "Mesh Attributes", and deactivate zones 1 and 2; zone 1 is the symmetry plane (BC#1, reflection) and zone 2 is the outer box (BC#3, characteristic inflow/outflow). This should leave zone 3 (BC#4, viscous surface), which is the ATF itself. Now, uncheck "Mesh" and "Boundary" and check "Contour" in the main Tecplot window, and you should see a plot of pressure coefficient (Cp) on the ATF surface. From this point, you could adjust the contour colors and levels, mirror the solution about the X-Z plane (to produce a full aircraft), and use numerous other capabilities in Tecplot. These are left up to the user.



Visualizing surface Cp contours in Amtec Tecplot

Summary

We hope this document has provided a useful and informative tour of TetrUSS on Mac OS X! For additional training in TetrUSS, please see the "Training and Support" document. Due to limited resources, we can only provide tech support to users who have taken the TetrUSS training class, and that includes tech support for this "Getting Started" tour. However, please contact the author by e-mail if you feel something is in error, if you have a suggestion/comment, or if you encounter a bug with the software. Please note that we are unable to answer basic questions regarding the use of OS X, the terminal, or Tecplot.